

Evolution of a Compact TOF Mass Spectrometer from Space Exploration to the Internet of Things

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technology

2019 HEMS Workshop

The background of the slide features a large, detailed image of the planet Jupiter, showing its characteristic bands of white, orange, and brown. In the lower right portion of the image, the moon Io is visible as a small, dark sphere. The overall scene is set against a dark, starry space background.

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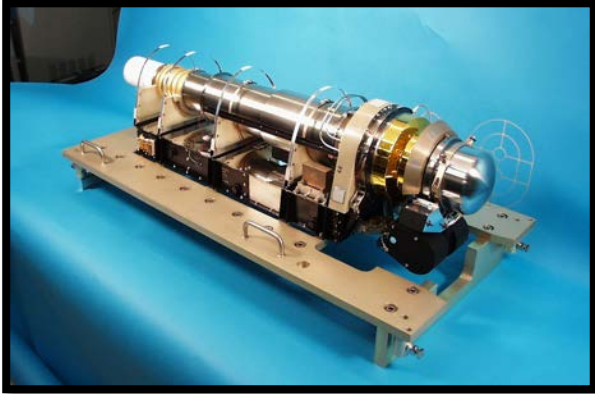
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Part 1: Neutral and Ion Mass Spectrometer (NIM)

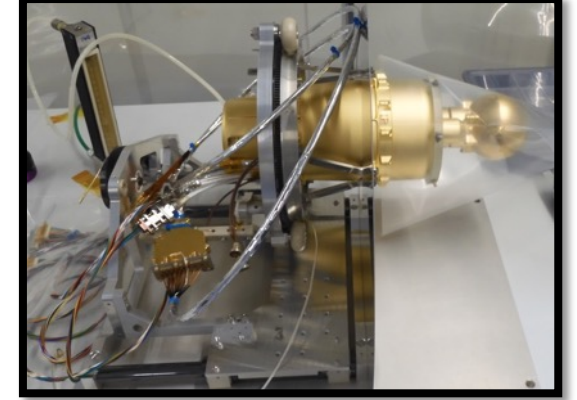
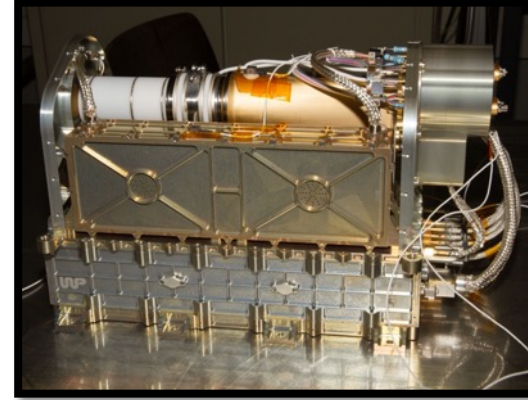
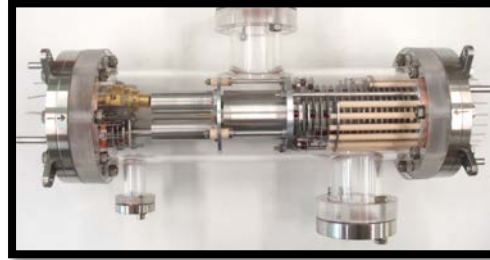
D. Lasi, M. Tulej, S. Meyer, M. Fohn, R. Fausch, D. Piazza, M. Gerber, S. Brungger, G. Bruno, M. Rieder, M. Lüthi, A. Nentwig, S. Oeschger, P. Fahrner, M. Althaus, P. Gubler., M. Gruber, H. Munz, and P. Wurz

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4 Generations of TOF Space Mass Spectrometers



~1m



~30 cm

- | | | |
|-------------|------------------|---------------------------------|
| ■ 1998-2002 | RTOF | Rosetta (European Space Agency) |
| ■ 2015 | EGT | P-BACE (Balloon mission) |
| ■ 2011-2018 | NGMS | Luna-Resurs (Roscosmos) |
| ■ 2012-Now | <u>NIM (PEP)</u> | JUICE (European Space Agency) |

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The Neutral and Ion Mass spectrometer (NIM)

Mission profile of JUICE (2022)

- ❖ Flybys Europa, Ganymede, Callisto
- ❖ Orbit around Ganymede

Key goals of the mass spectrometer

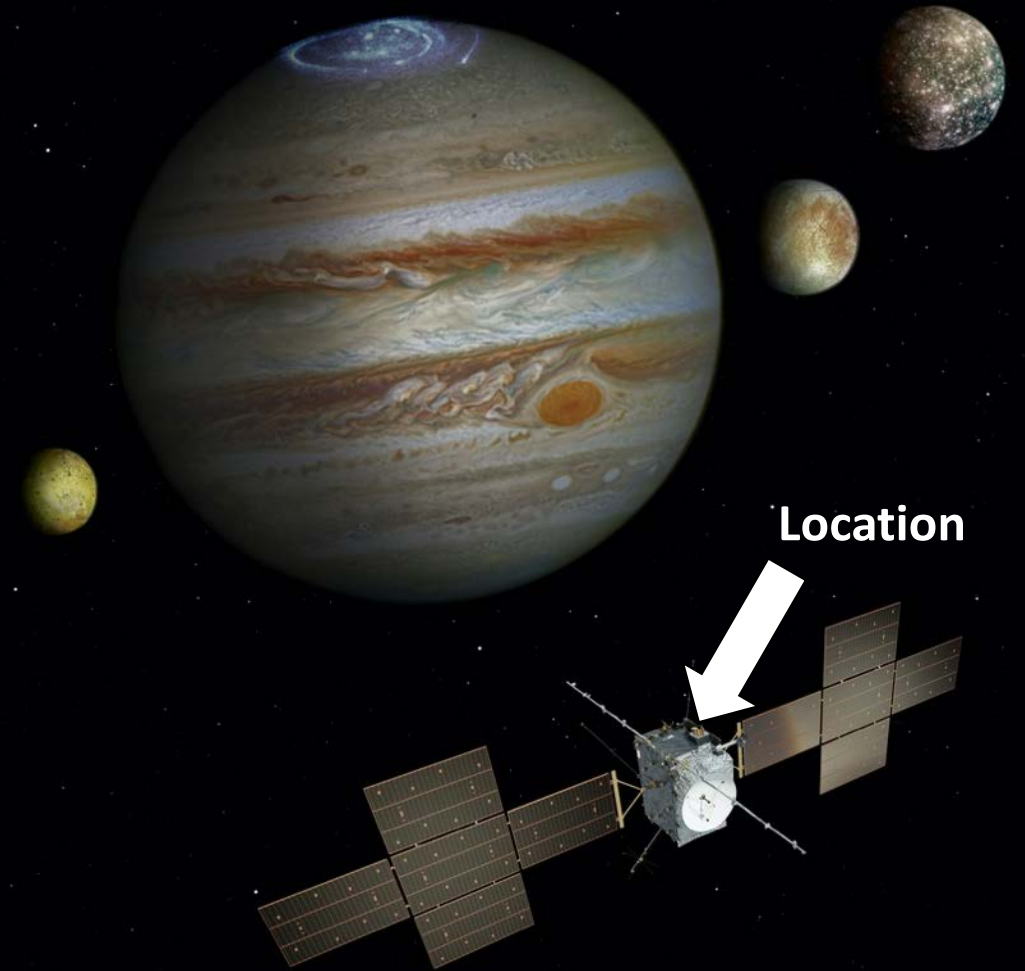
- ❖ Characterize exosphere of icy moons (neutral, ions)
- ❖ Achieve a spatial resolution of ~25–50 km

Key instrument requirements

- ❖ Mass range up to 300 u
- ❖ Mass resolution $> 500 M/\Delta M$
- ❖ Dynamic range $> 10^5$ (5s integration)
- ❖ Measurement cadence 5s (flyby), 1s (plume)

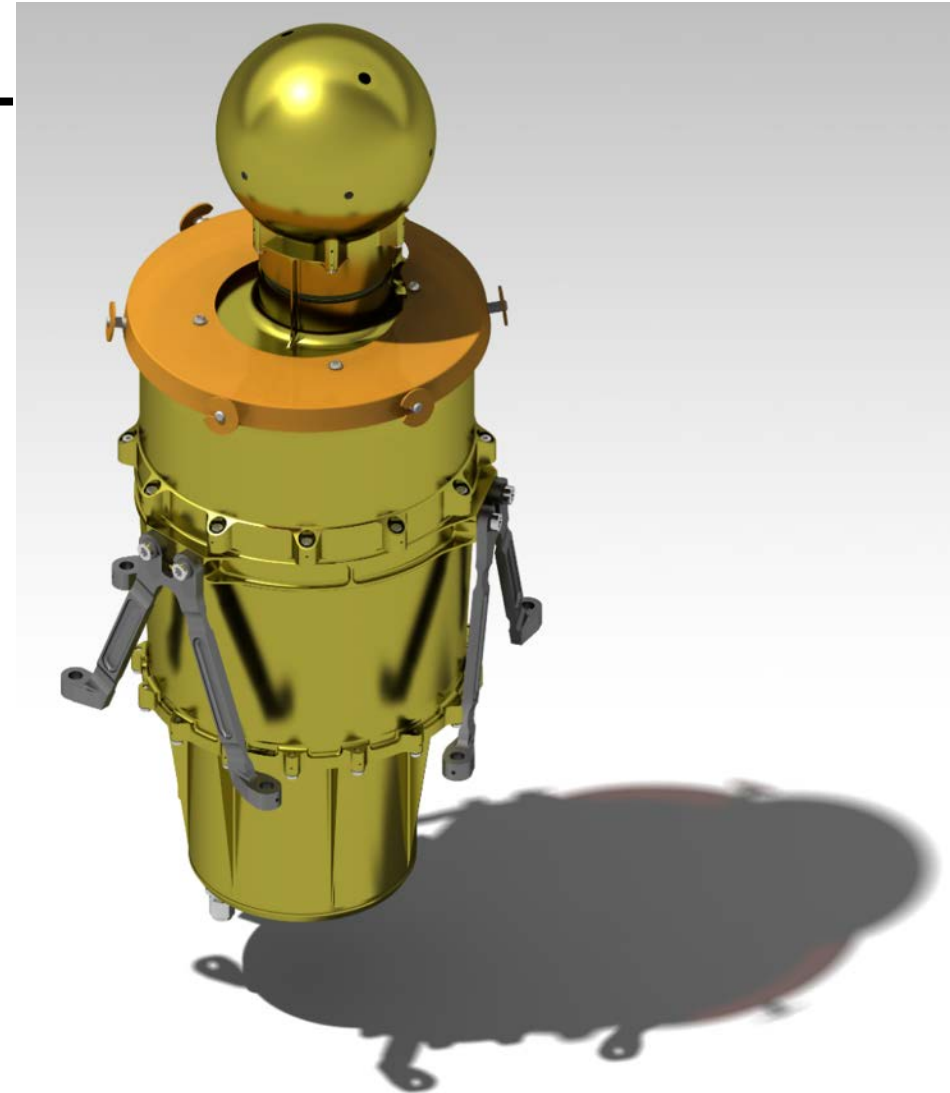
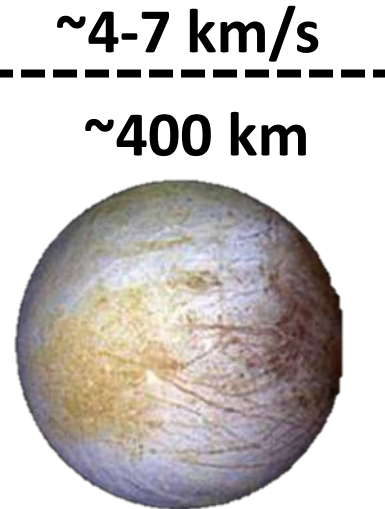
Constraints

- ❖ Survive and measure in the radiation environment (@Europa)
- ❖ Mass 3 kg (excluding radiation shielding)
- ❖ Power 12 W (before common DCC)

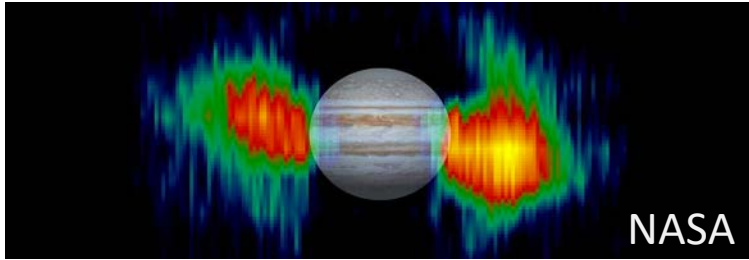


Instrument Overview

- **Architecture:**
 - Reflectron TOF MS
 - Electron impact ionization
 - MCP detector
 - ~10 kHz repetition rate
- **Three different operation modes**
 - **Neutral:** direct FoV + ionization
 - **Ion:** direct FoV (no ionization)
 - **Thermal:** antechamber + ionization (2pi FoV + extended energy range)
- **Switching mechanism (shutter)**

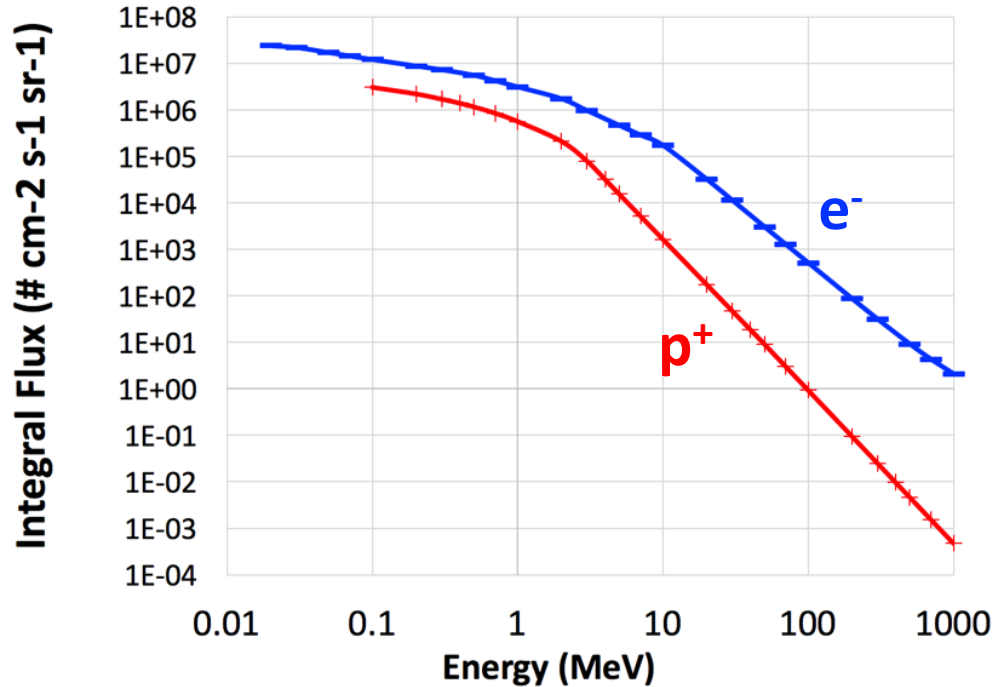


A Very Harsh Radiation Environment!



→ Strategy: shield differently against fluence and flux

Worst-Case (Europa)



Shield electronics against fluence (TID)

- Driver end of life
- Goal <50 krad Si
- Need ~2 mm Ta
- How? PEP vault

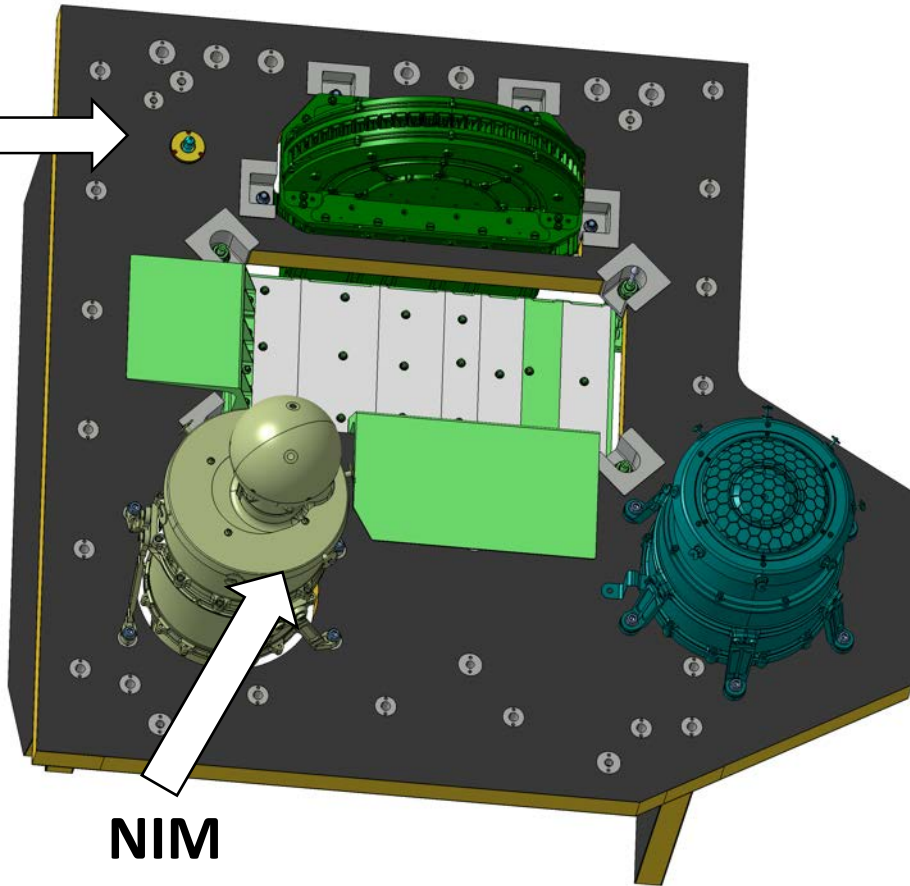
Shield detector against flux

- Driver worst case
- Goal <~10⁵ #/s
- Need ~10 mm Ta
- How? Dedicated shield

Total dose (unshielded: ~0.05 mm Al) → ~160 Mrad

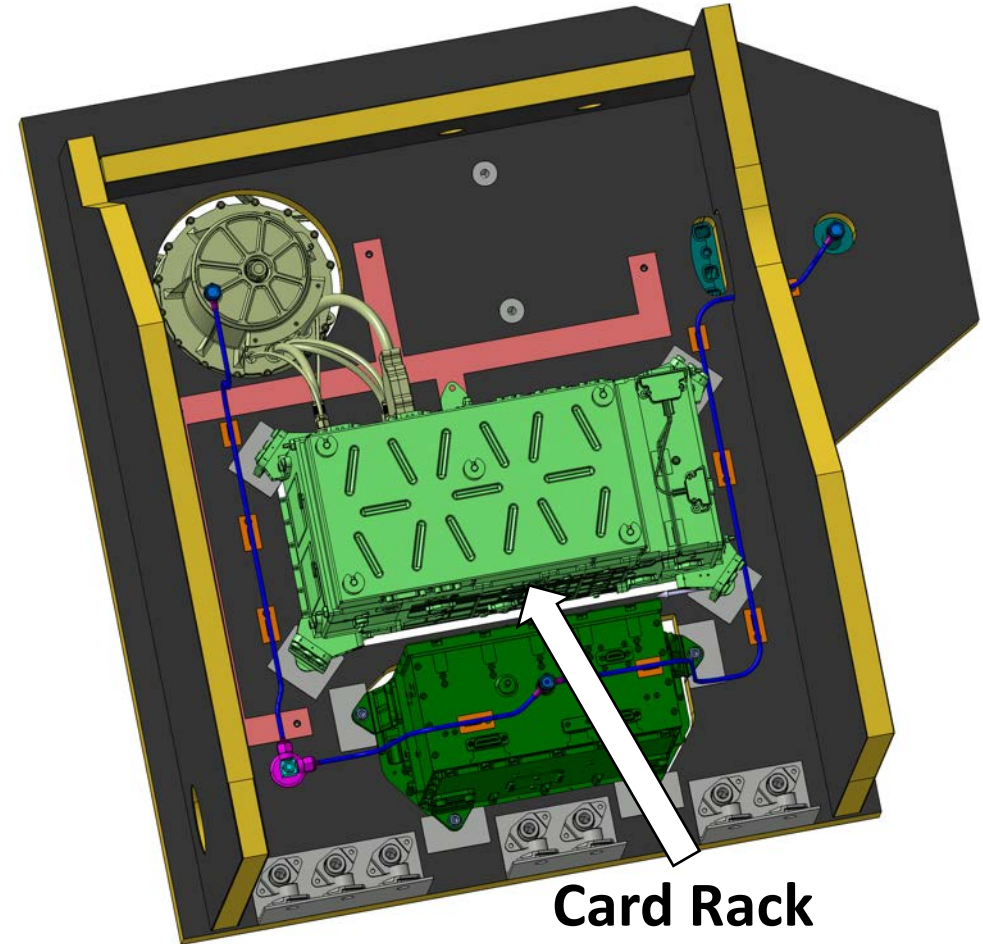
Particle Environment Package (PEP) Nadir Unit (NU)

Spacecraft
CFRP
panel



NIM

Tiny detector inside, with its shield
Everything else unshielded



Card Rack

'Common Vault' of PEP NU Electronics

2 mm WCu

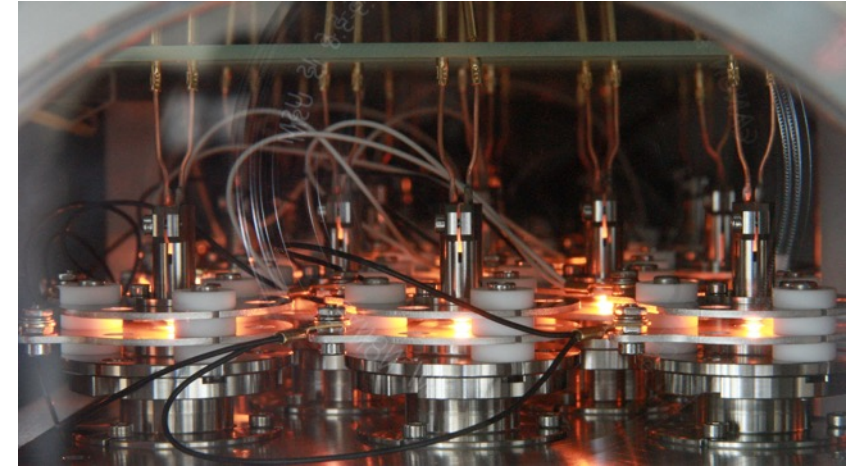
Coming Next → selected design insights 'following the ions'

Electron Emitters



- Considered **cold cathodes** (FEA, CNT)
 - Interesting because low power, but
 - low TRL and heritage, uncertain reliability, limited availability
- Decided for **thermionic emitters** from Kimball Physics
 - Enhanced yttria oxide (baseline)
 - Barium oxide (backup)
- **AC instead of DC to save power!**

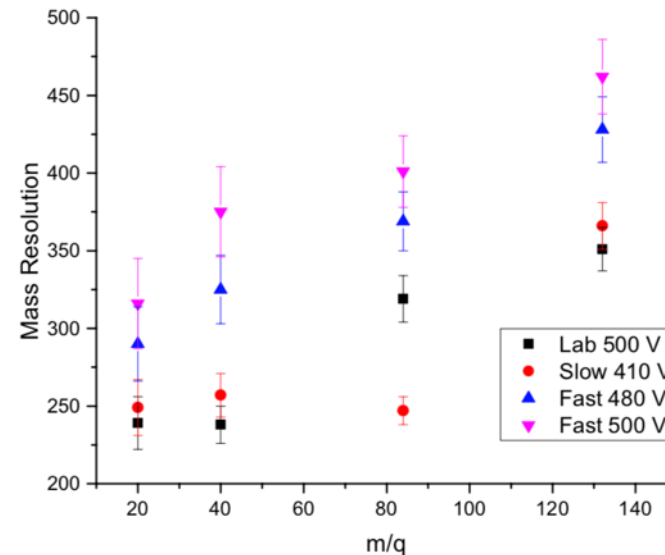
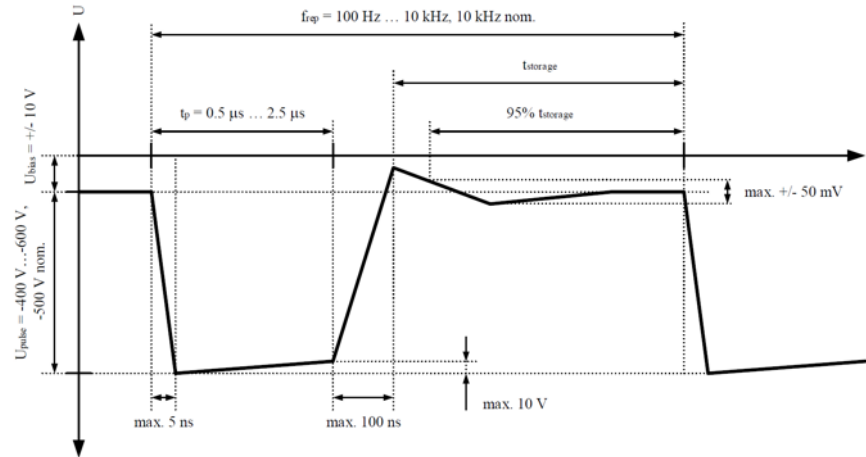
Property	Y ₂ O ₃ (enhanced)
Lifetime	>10'000 cycles
Power at 300 μA emission	1.2 W



10,000 hours goal
~2,000 hours run so far without issues

Extraction with Fast High Voltage Pulser

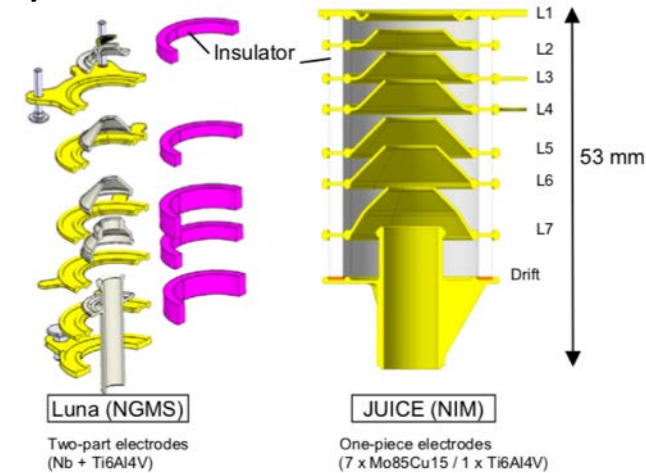
- **Critical requirements** →
 - Mass resolution $\sim S/N$
- **Limited choice of radhard MOSFETs**
- **Prototyped two pulser designs:**
 - 2 MOSFETs in cascade
 - Can go up to 600V (derating)
 - 'Slower' (~ 6.5 ns)
 - 1 MOSFET
 - Can go up to 480V (derating)
 - 'Faster' (4.5 ns)
- **Selected: faster pulser**
 - Takeaway: leading edge steepness more important than pulse height!
- **Optimized shape at point of application** with optimized ceramic electrode



Integrated ceramic pull grid



- **Heritage:** RTOF (Rosetta), NGMS (Luna-Resurs)
- **Ion source**
 - New one-piece electrode design
 - From Nb + Mo to MoCu (85:15)
 - Better manufacturability
 - Same CTE than alumina
 - See Elsener et al., 2019

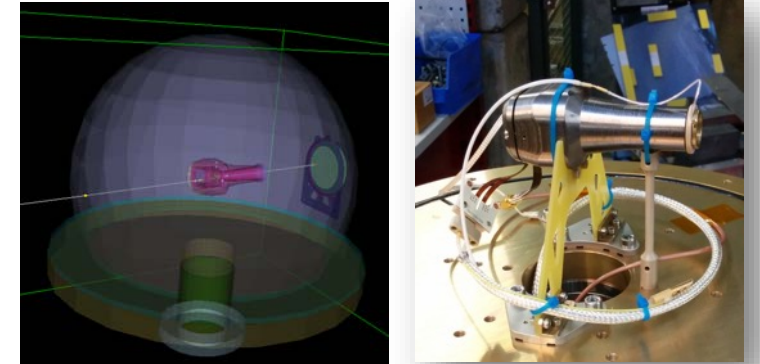
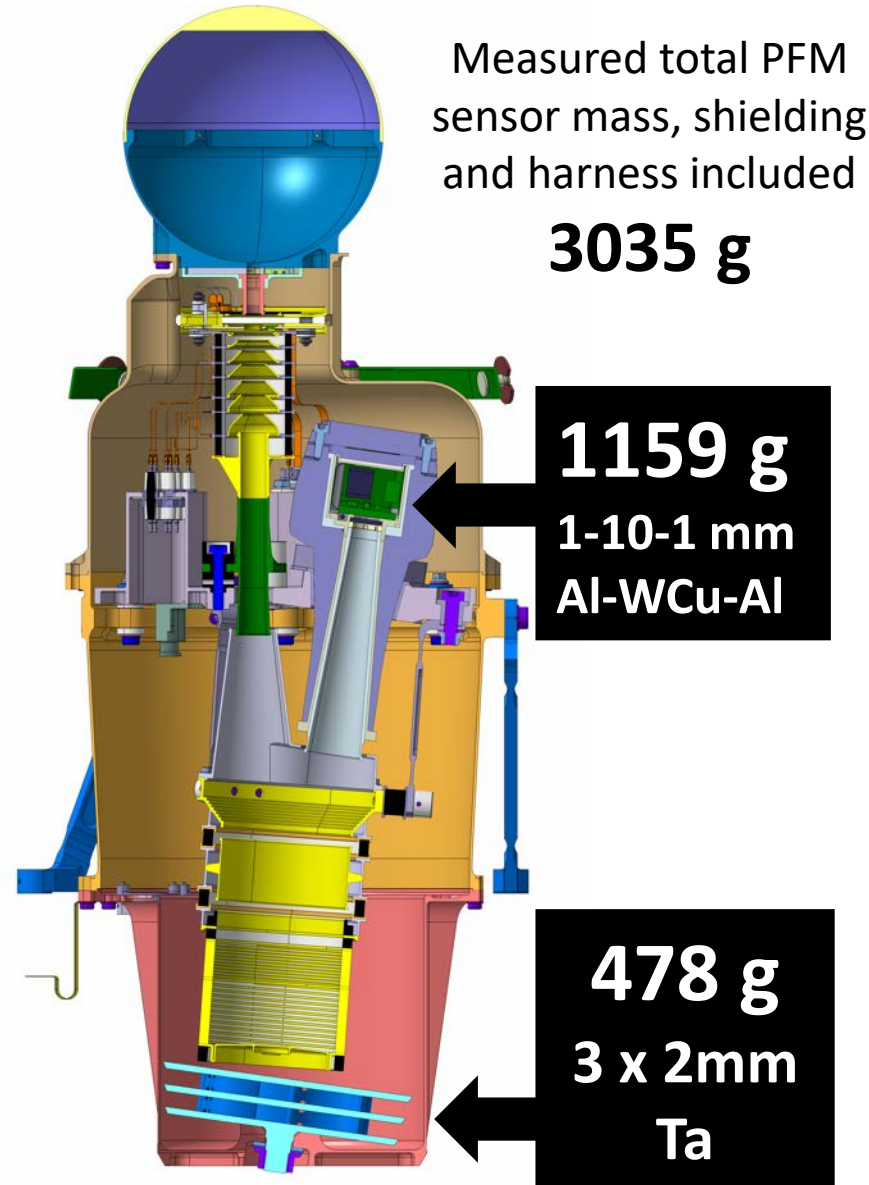
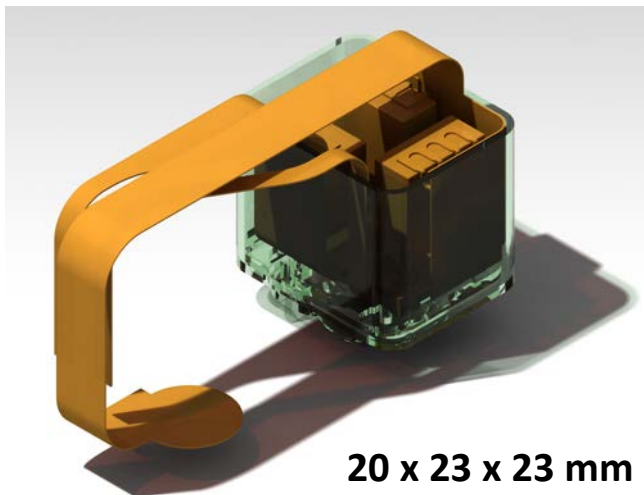


- **Reflectron**
 - Screen-printed electrodes (DuPont Series 2000)
 - Excellent mechanical properties
 - Min. field corrugation → Efficient use of volume
 - Verified with 18 MeV p + up to 85 Mrad
 - See Scherer et al., 2006; Lasi et al., 2017

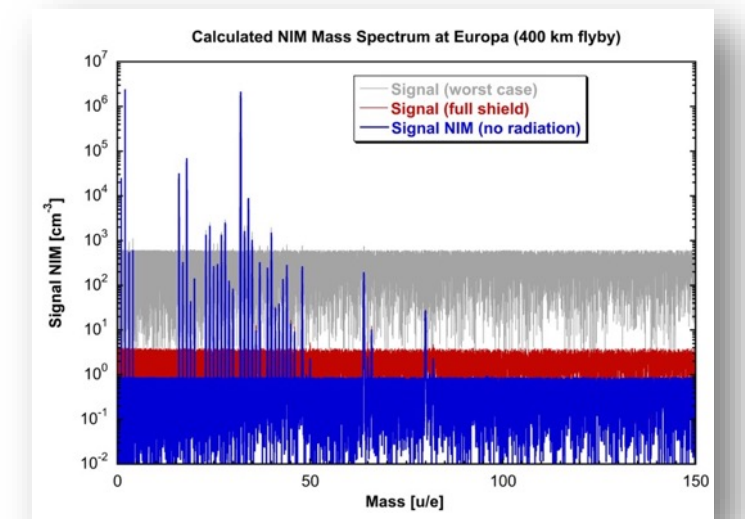


Micro-Channel Plate (MCP) Detector and Shielding

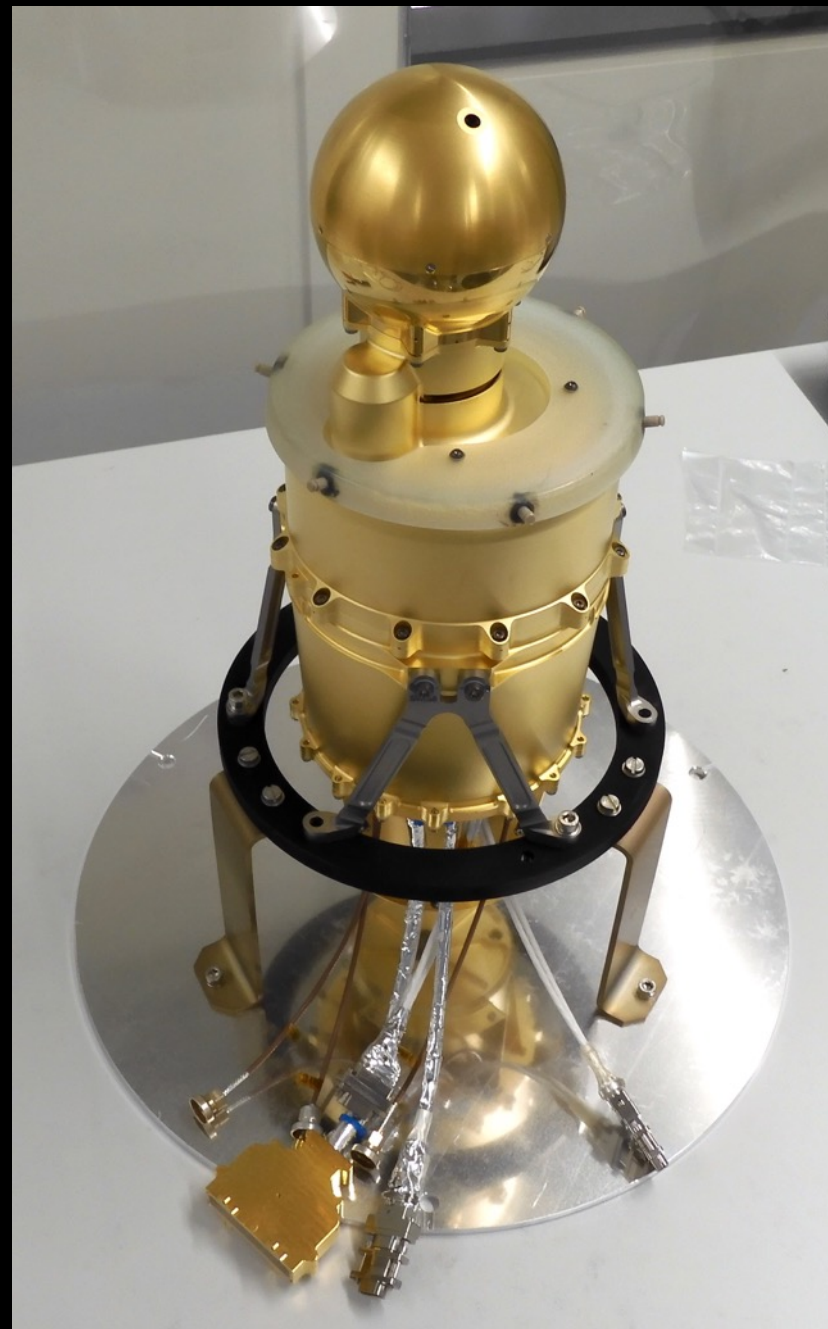
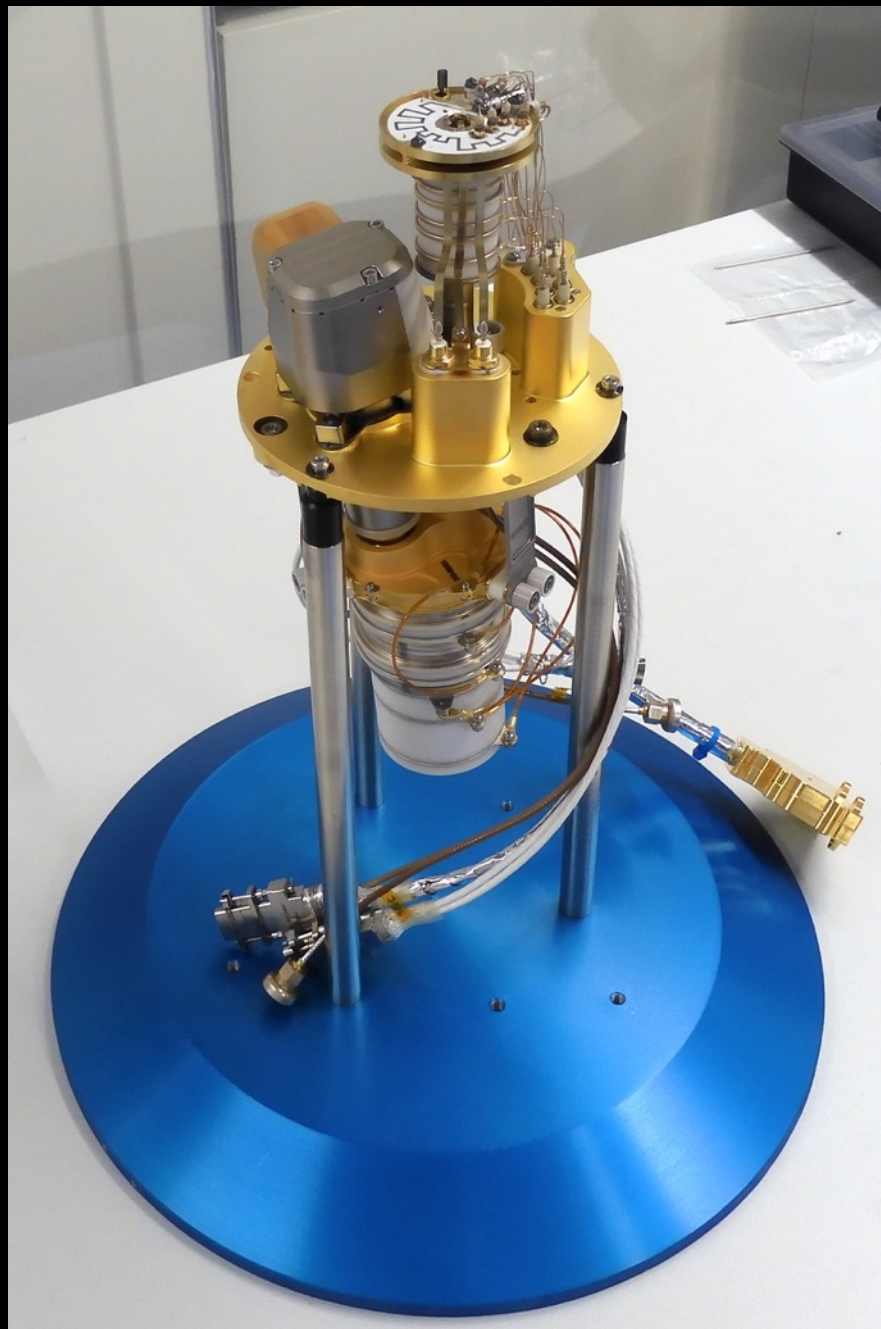
- **10 mm MCPs**
- **Design drivers**
 - Minimum volume
 - Availability of EEE parts
 - Signal quality
 - Routing through shield
- **Rigid-flex PCB**
- **Folding process**



- **Geant4 sim.** (PSI, IRF, Aberystwyth)
- **Tests with 11-345 MeV e⁻** (PSI)
- See Tulej et al., 2015, 2017 and Lasi et al., 2019



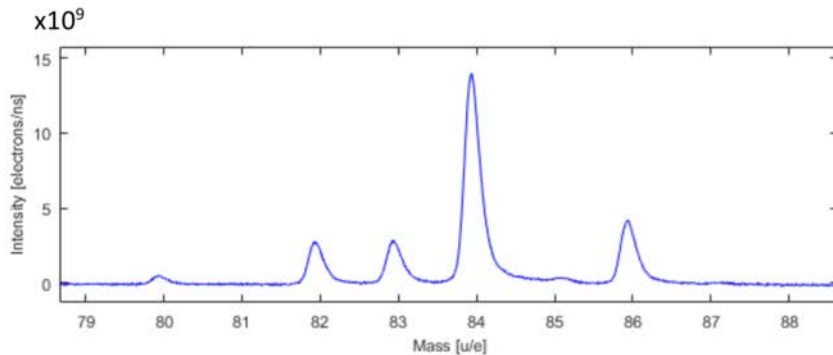
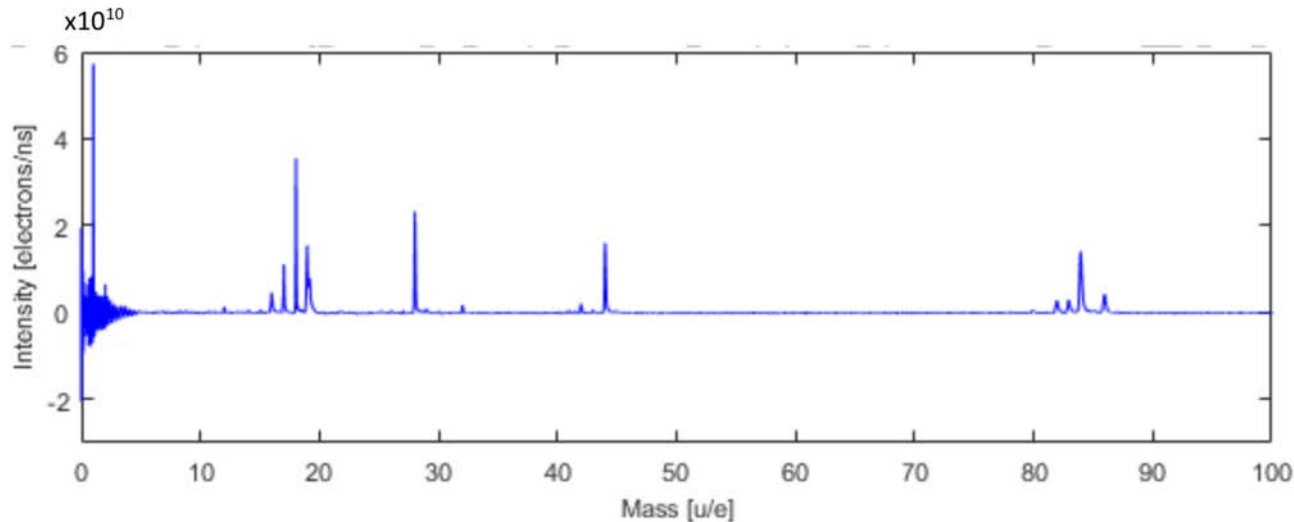
NIM Sensor
PFM (2019)



First Result

SNR: 301 $P_{BG} = 1.5 \cdot 10^{-9}$ mbar $U_{MCP} = 1.75$ kV

Mass Res Kr84: 347 +-19 $P_{tot} = 4 \cdot 10^{-9}$ mbar



- ✓ Very 1st spectrum!
- ✓ September 9th, 2019
- ✓ With initial manual settings
- ✓ With lab electronics (but our own pulser)
- ❑ Now optimizing using Particle Swarm Algorithm
- ❑ Connecting to flight electronics in Jan 2020

References

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4. S. Meyer, et al., "Mass spectrometry of planetary exospheres at high relative velocity: direct comparison of open- and closed source measurements," Geosci. Instr. Method. Data Syst. 6(1), (2017) 1-8, doi:10.5194/gi-2016-28.
5. M. Tulej, et al., "Experimental investigation of the radiation shielding efficiency of a MCP detector in the radiation environment near Jupiter's moon Europa," Nucl. Instr. Meth. B383 (2016), 21-37, DOI: doi:10.1016/j.nimb.2016.06.008.
6. M. Tulej, et al., "Detection efficiency of microchannel plates for e- and π - in the momentum range from 17.5 to 345 MeV/c," Rev. Sci. Instr. 86 (2015) 083310, 1-12, doi: 10.1063/1.4928063.
7. S. Barabash, et al., "Radiation mitigation in the Particle Environment Package (PEP) sensors for the JUICE mission," European Planetary Science Congress 2015, EPSC Abstracts, Vol. 10, id. EPSC2015-589.
8. S. Scherer, et al., "A novel principle for an ion mirror design in time-of-flight mass spectrometry," Int. Jou. Mass Spectr. 251 (2006) 73-81.

Part 2: From Space to IoT

D. Lasi^{1,2}, L. Hofer², J. Jost¹, A. Péteut¹, M. Gruber¹, S. Gasc¹

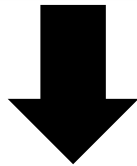
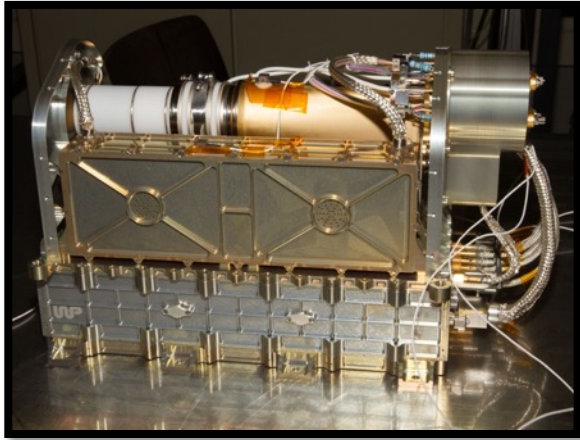
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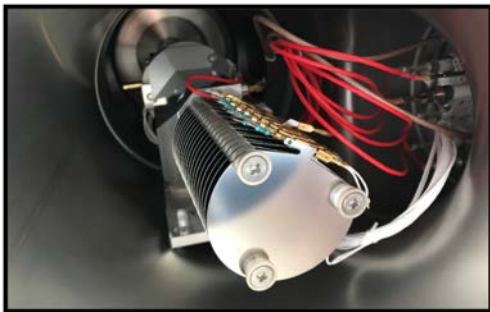


From Space to Earth

NGMS PFM (UBe, 2018)



Spacetek TOF and Electronics (2019)

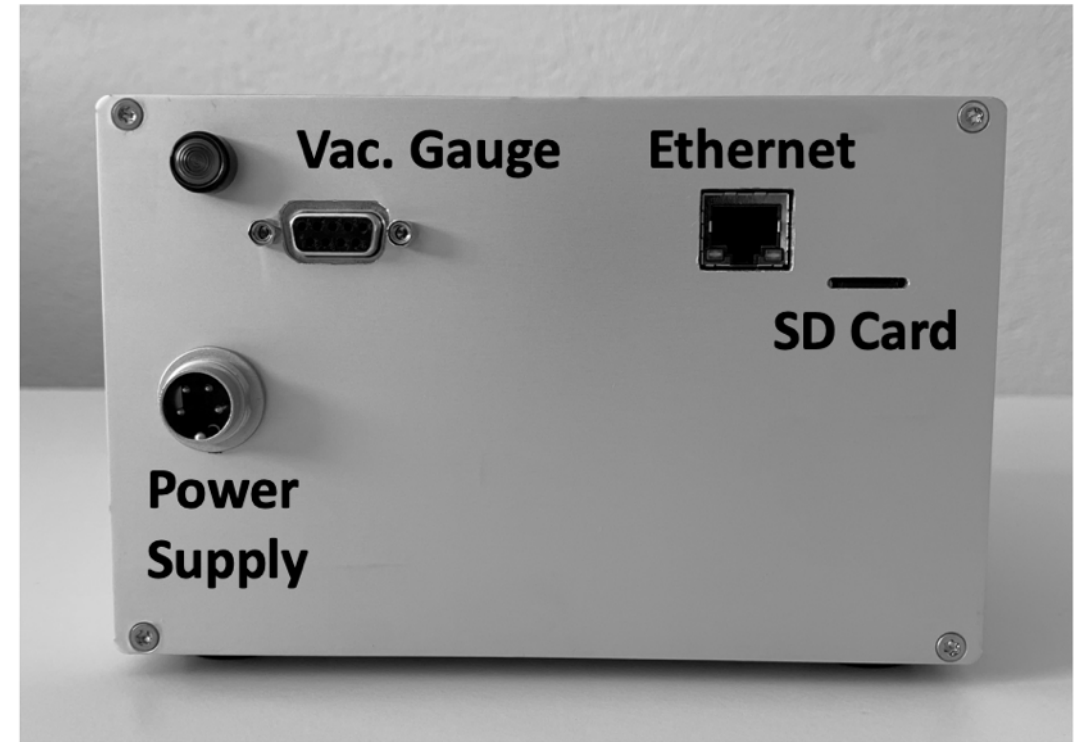
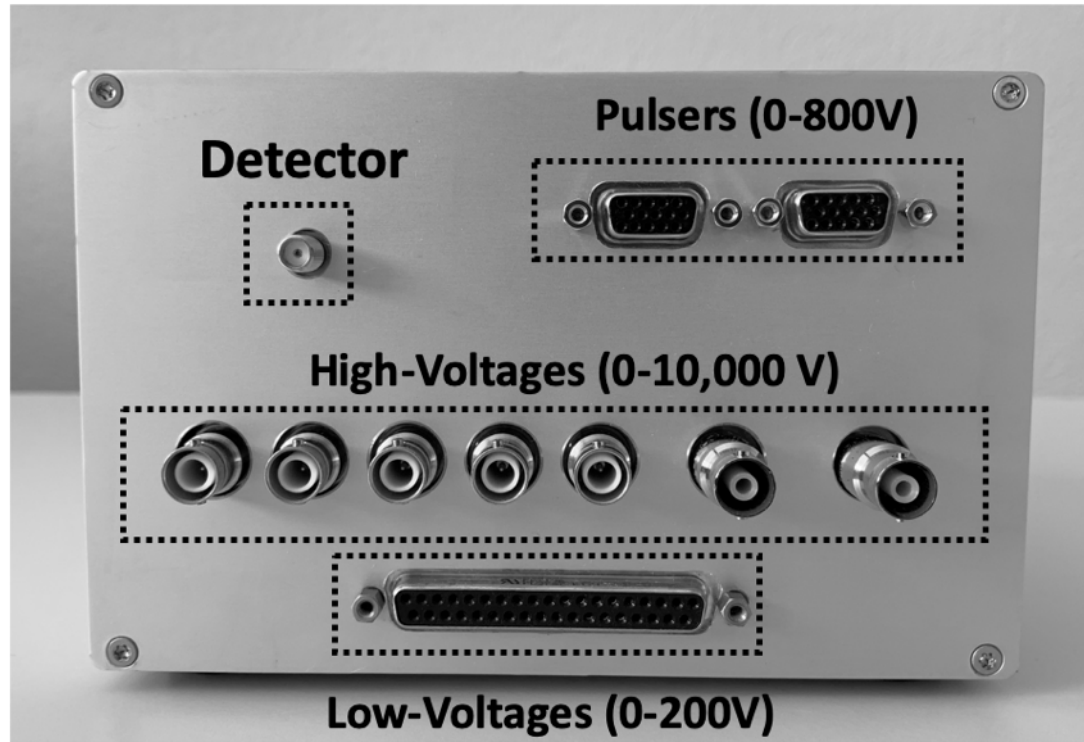


- **100% redesign from the ground up...**
- **What's IN?**
 - Most architectural decisions
 - Remote control and diagnostic
 - Automated particle swarm optimization
- **What's OUT? (for the time being)**
 - Metal-ceramic ion optics
 - MCP detectors (→CEM)
 - Vacuum electronics
 - Radhard design
(EEE, scrubbing, TMR, materials, shielding, ...)
- **What's NEW?**
 - Use of ARM System-on-Chip (SoC)
 - HTTP API (→ Internet of Things device)

Sensor length:
~300 mm

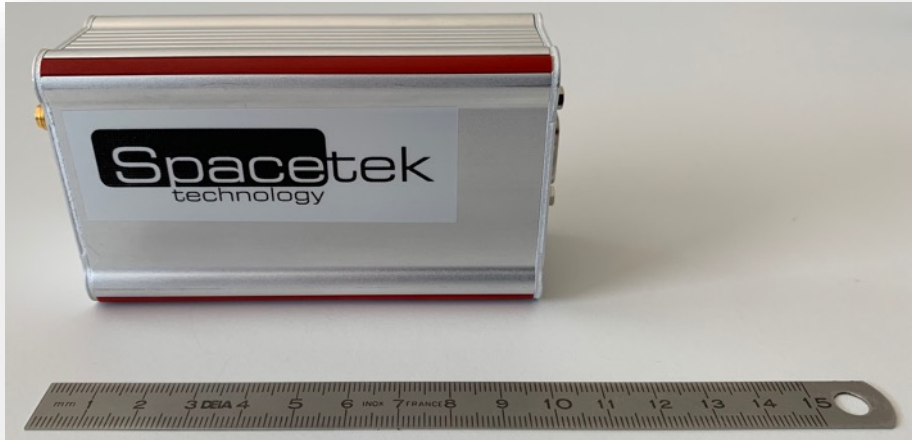
Optimized for
manufacturability
(assembly < ½ day)

Instrument Electronic Box

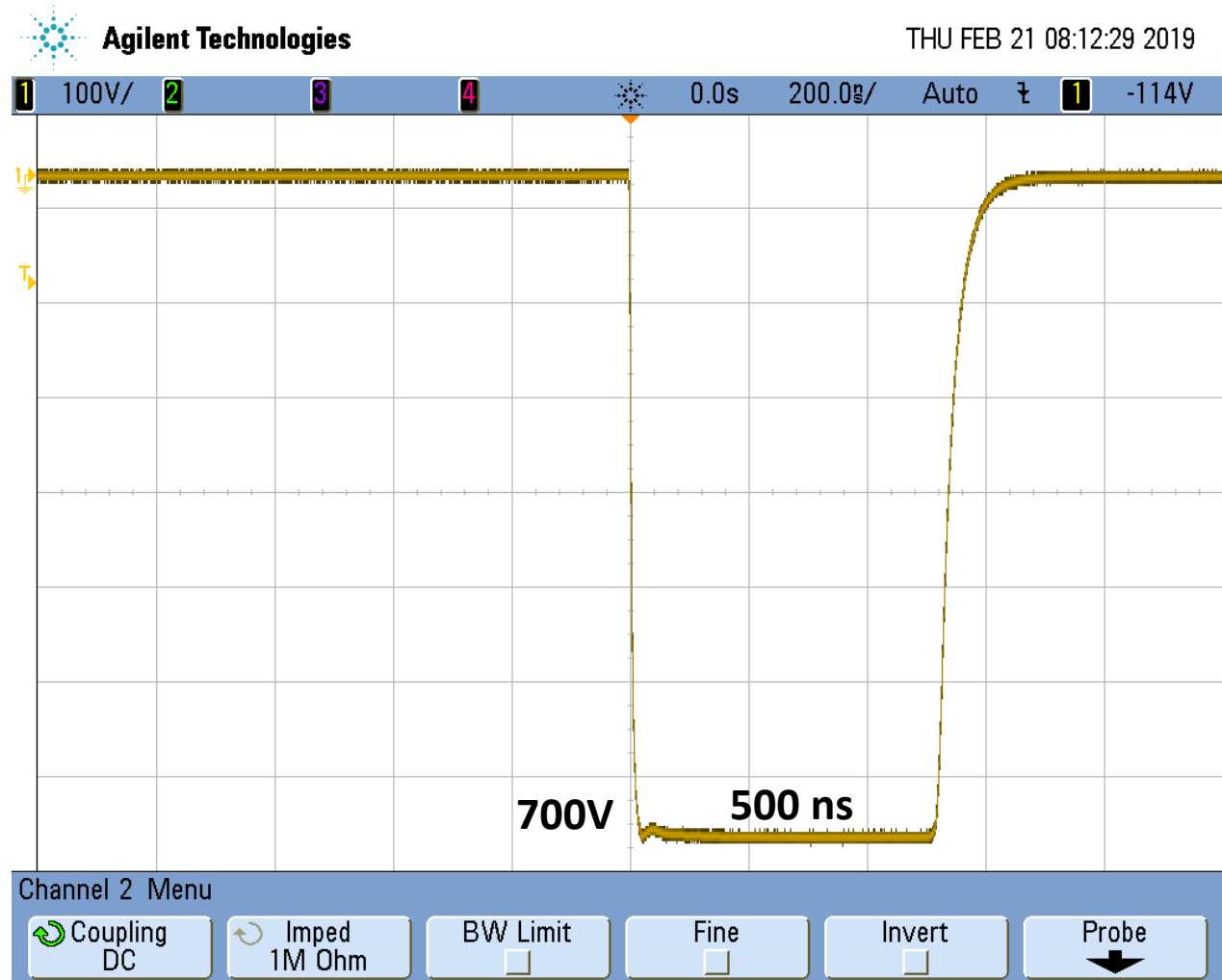


- **100% developed in-house**
- **Box size: 238 x 165 x 105 mm**
- **50% volume reduction possible with layout optimization**
- **Power consumption: ~25 W (excl. vacuum pump)**

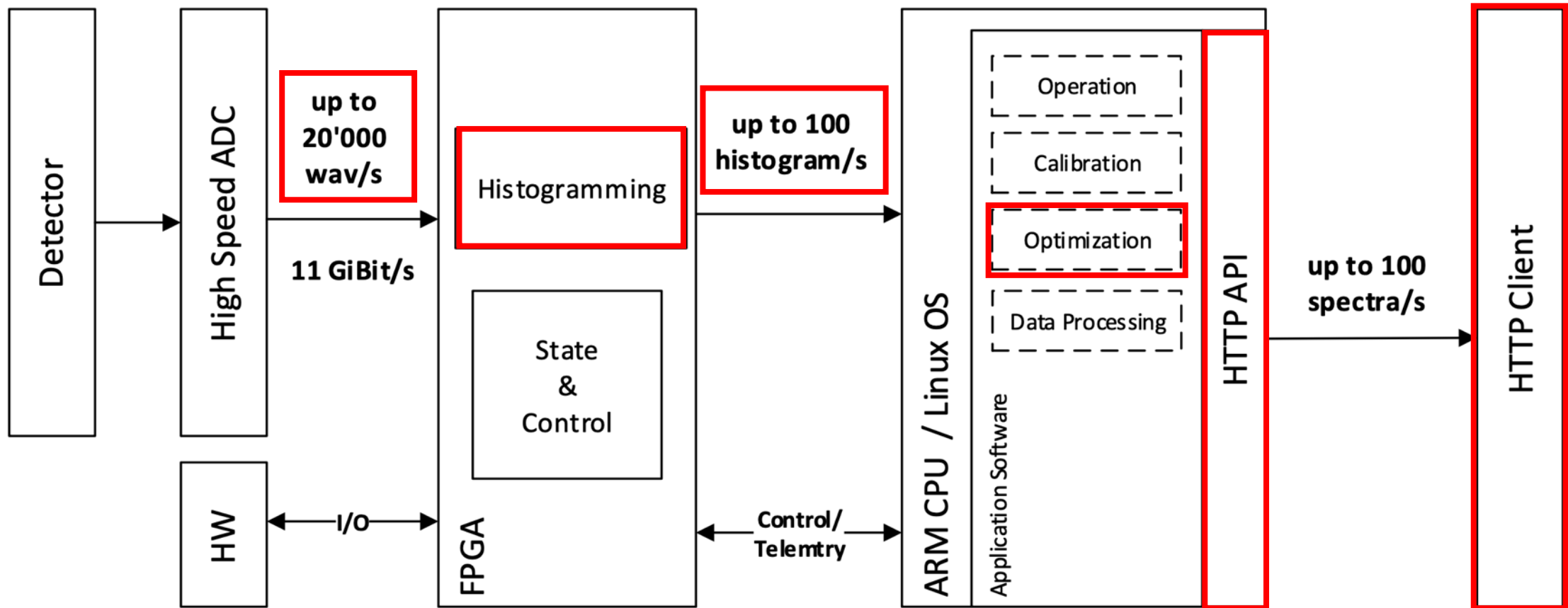
Fast High-Voltage Pulser



- **100% developed in-house**
- **External box:** flexibility, short cables
- **Pulse height:** up to ~800V (+ or -)
- **Adjustable duration**
- **Rise time:** ~2-3 ns



Electronics Design and Capabilities



First Results

Kr	10%	N2	10%
Xe	10%	Ar	60%
He	10%		

September 12th, 2019!

Integration time (full spectrum):

100 ms

(sum 1,000 single shots @ 10 kHz)

Mass resolution:

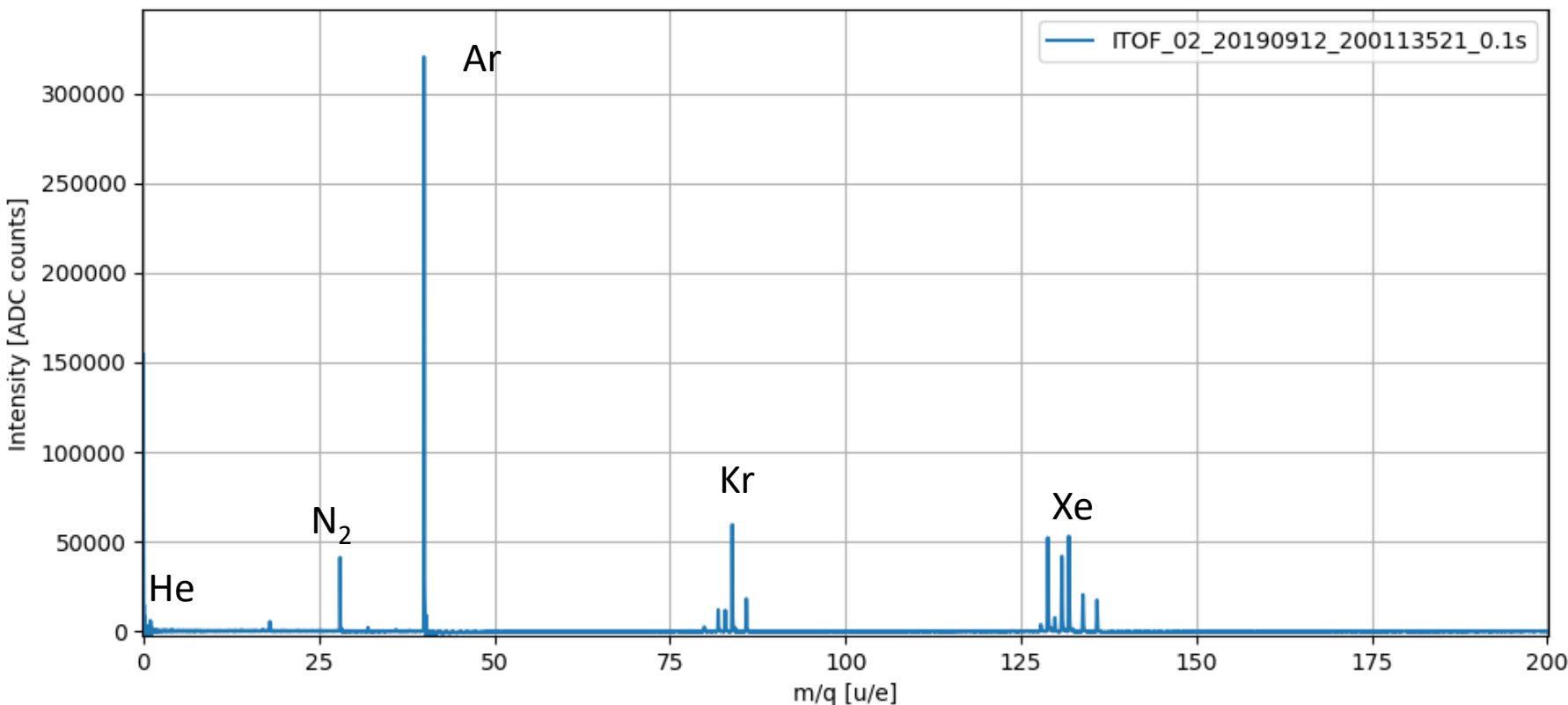
900 $M/\Delta M$ (CO_2)

1,250 $M/\Delta M$ (Xe)

Dynamic range

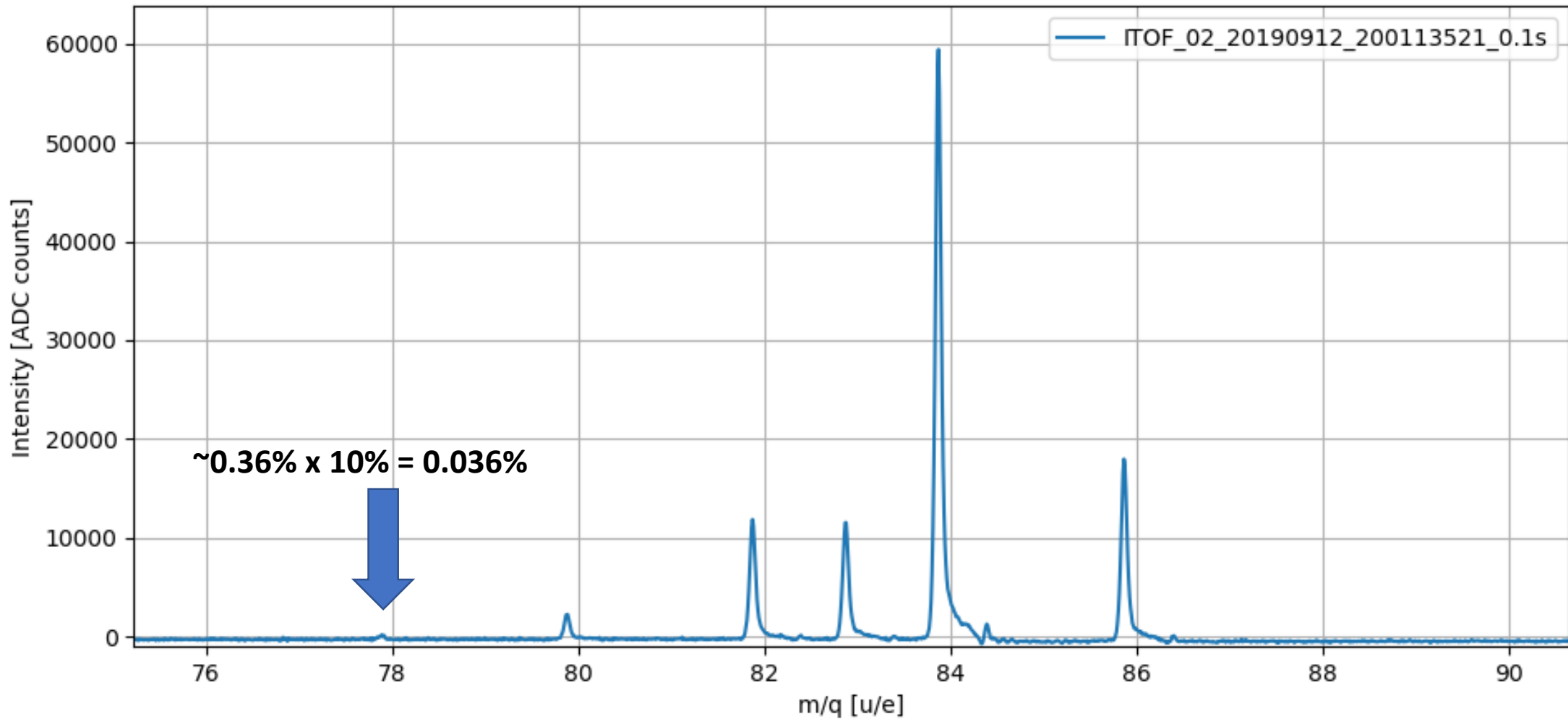
$\sim 10^5$

(six decades: achievable goal)

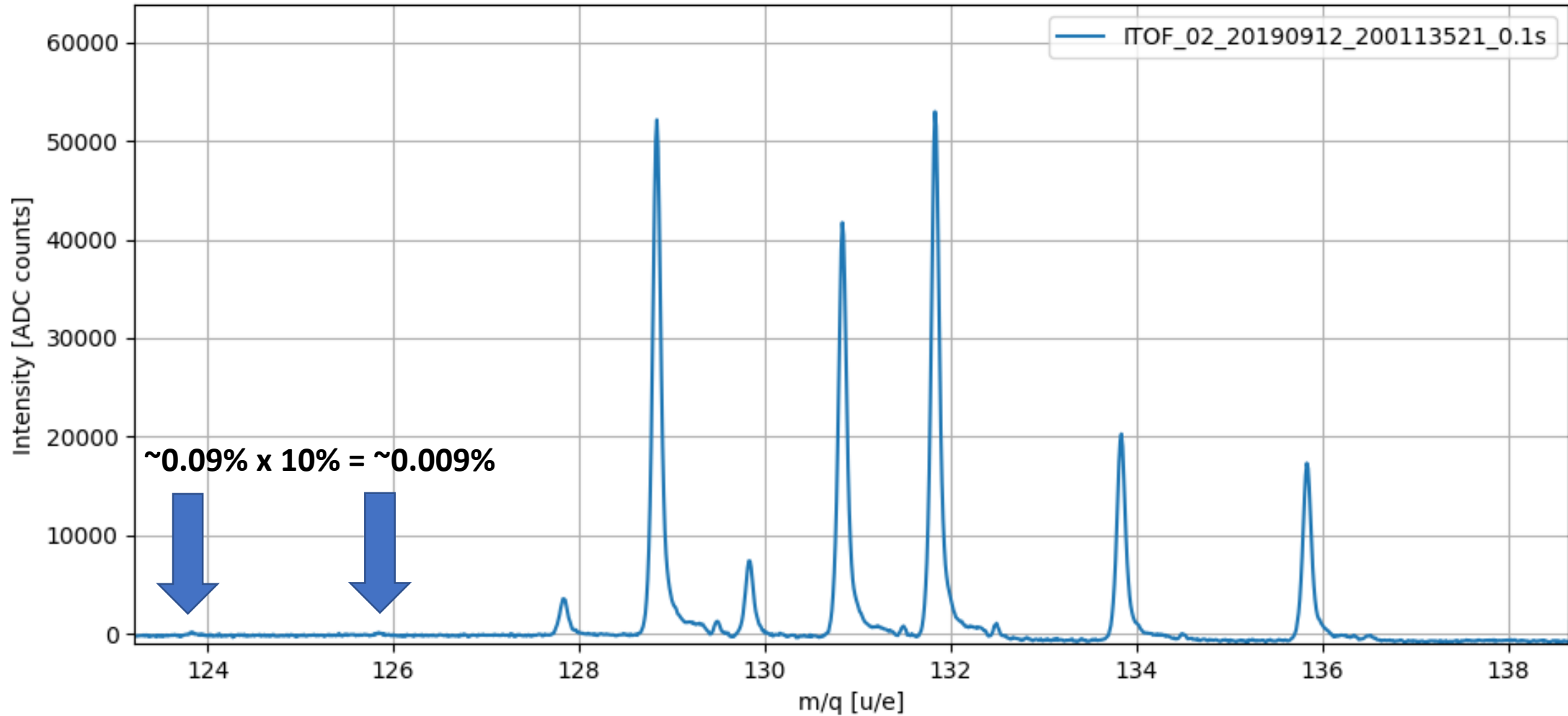


All the way up to 720 u

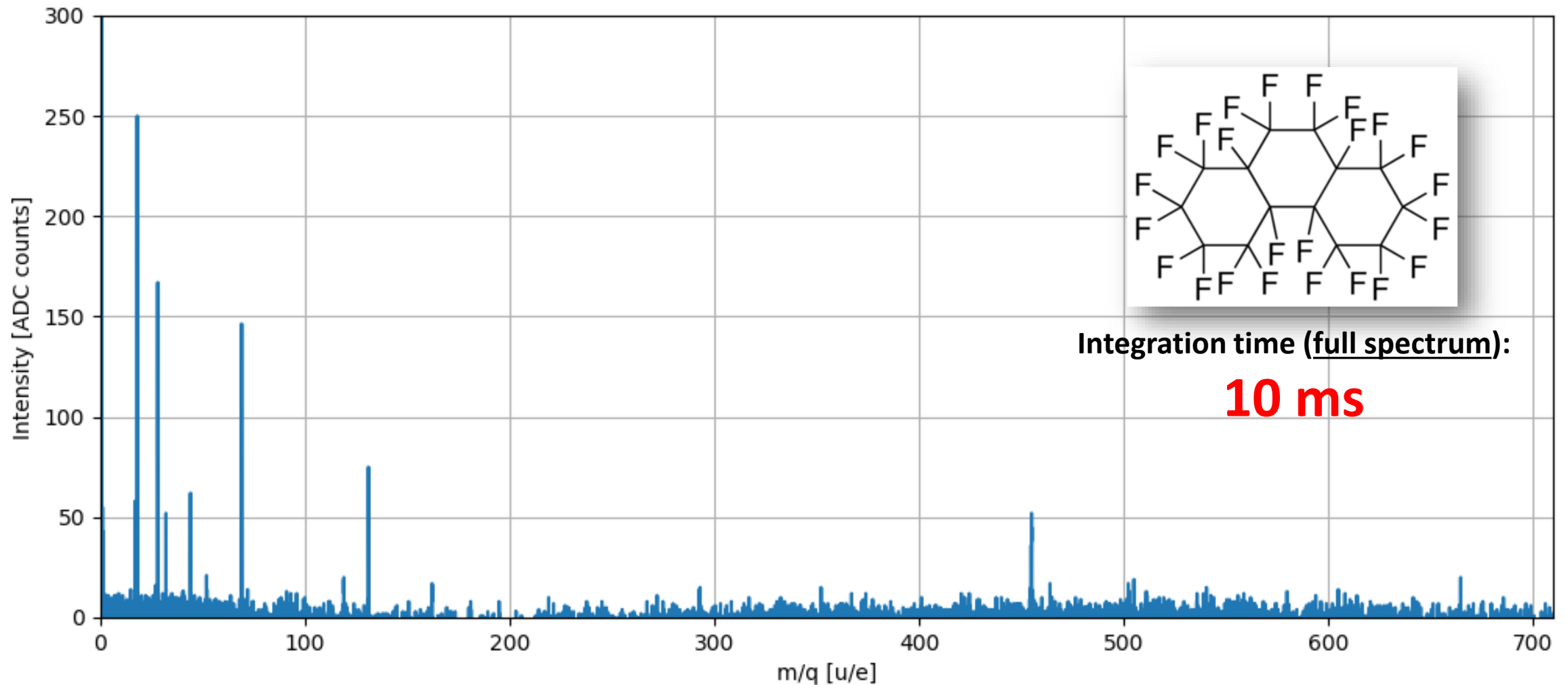
Zooming-in on the isotopes of Kr



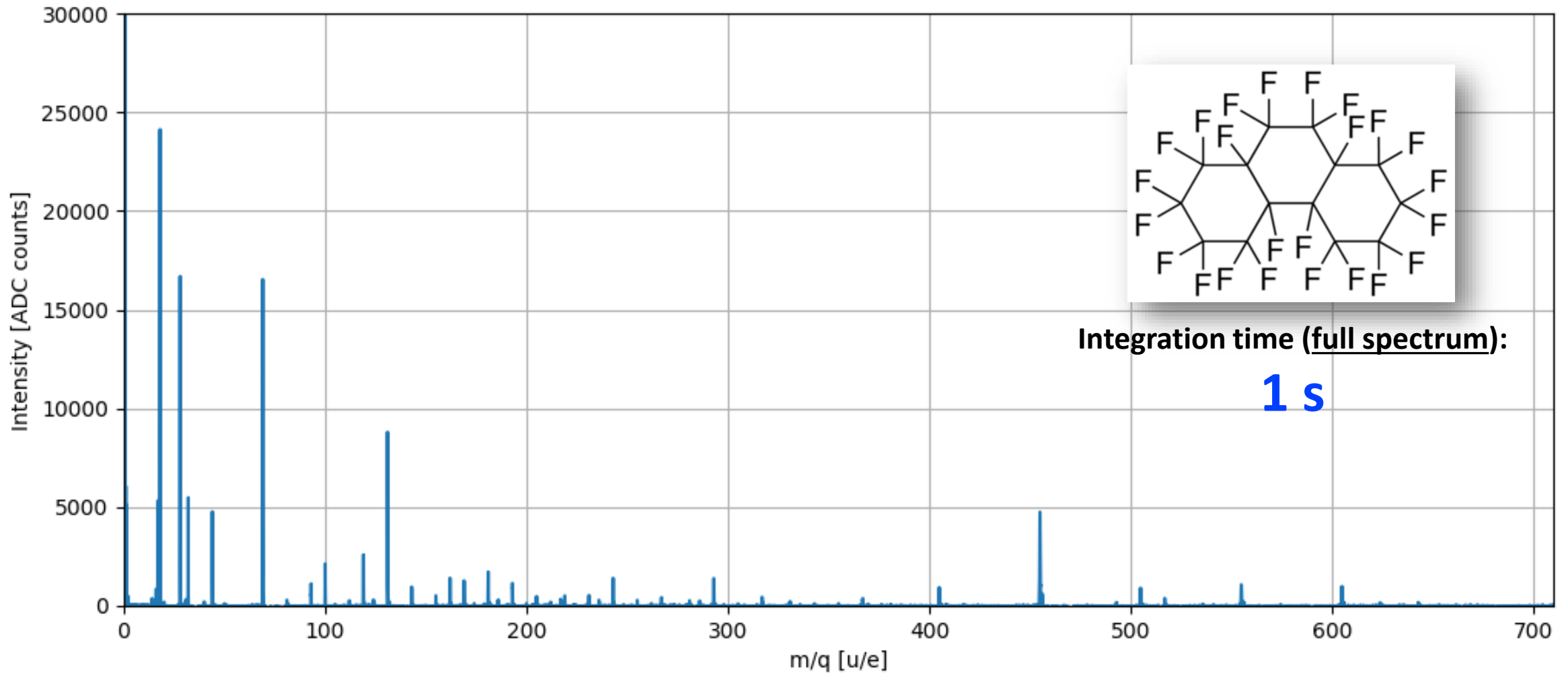
Zooming-in on the isotopes of Xe



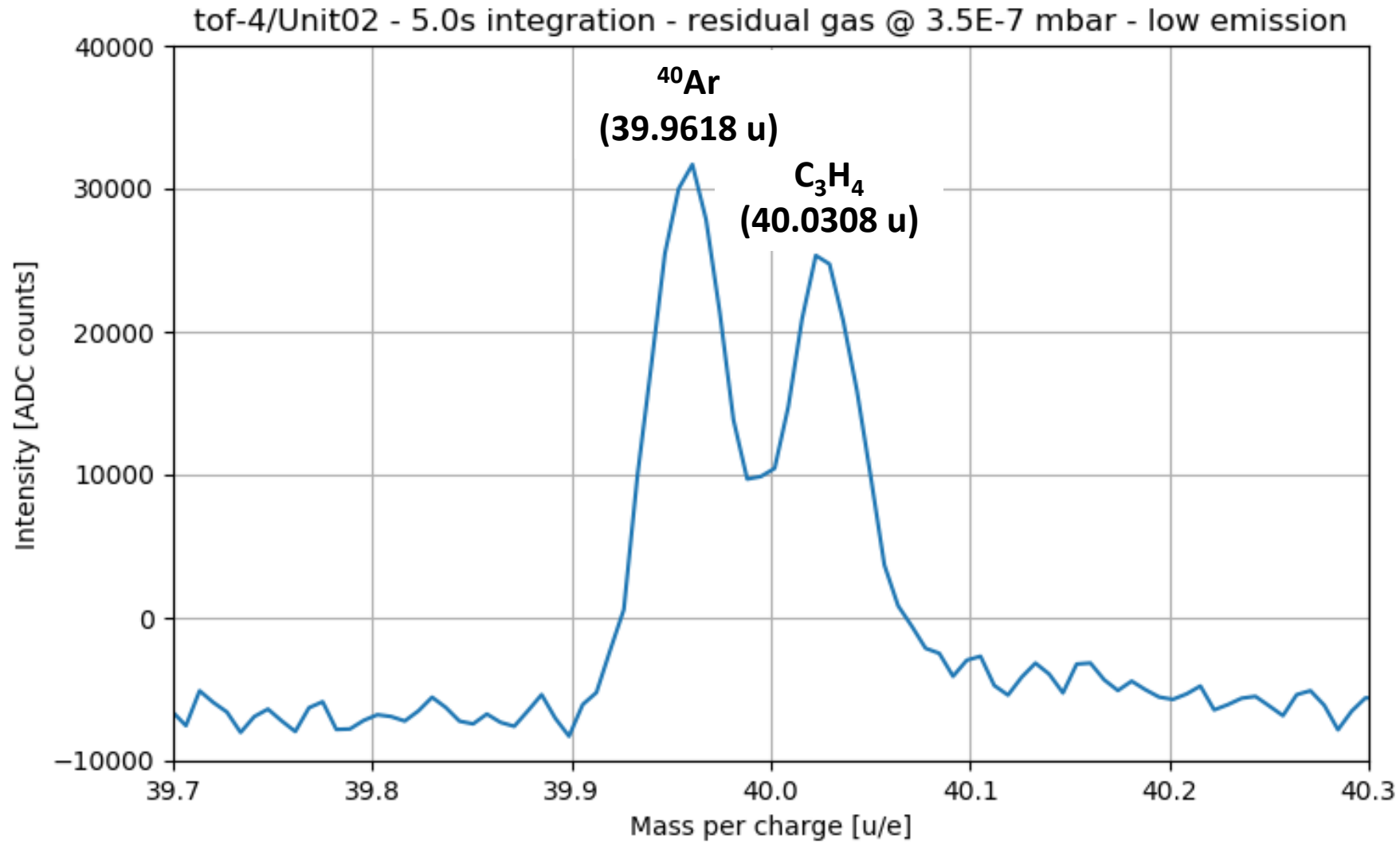
A Heavier Calibration Compound



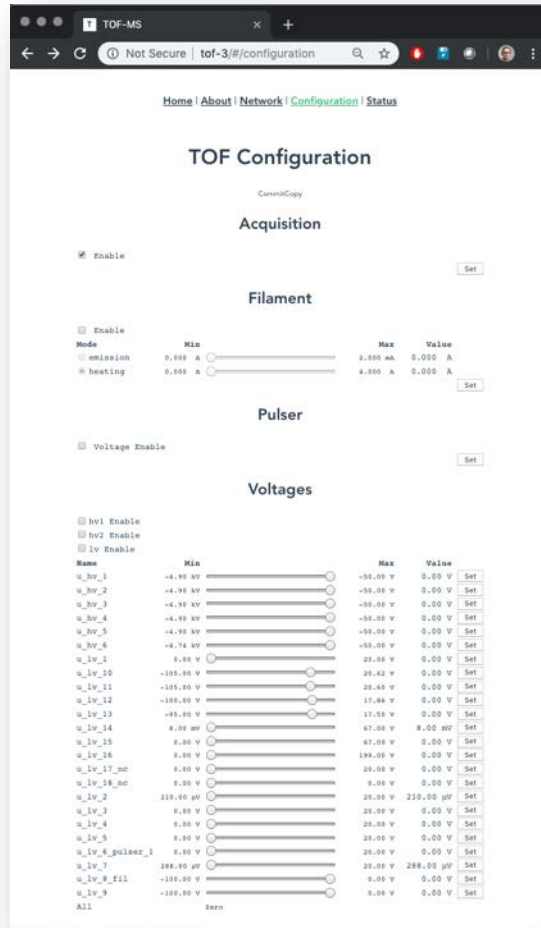
A Heavier Calibration Compound



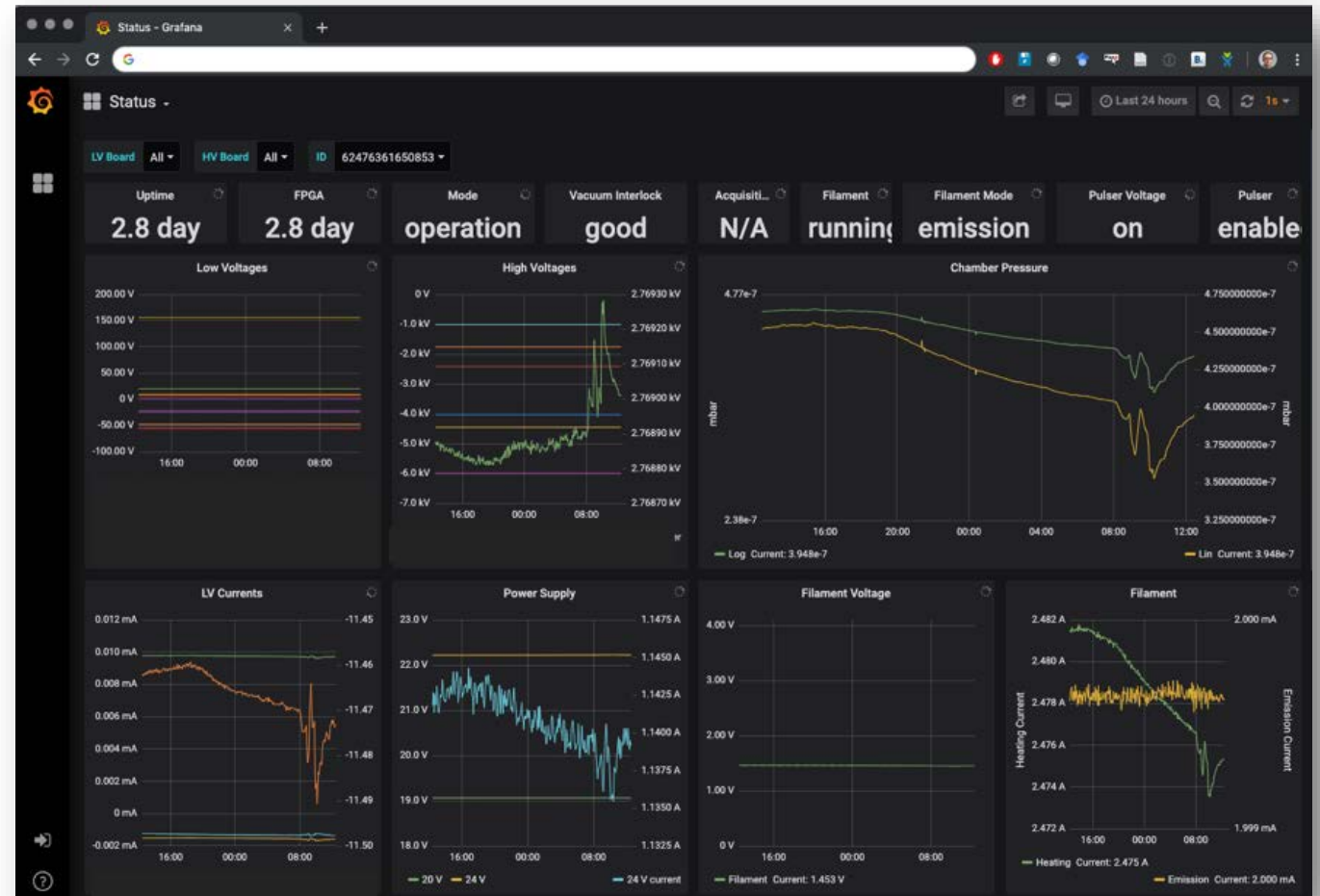
On Mass Resolution...



Full Remote Control

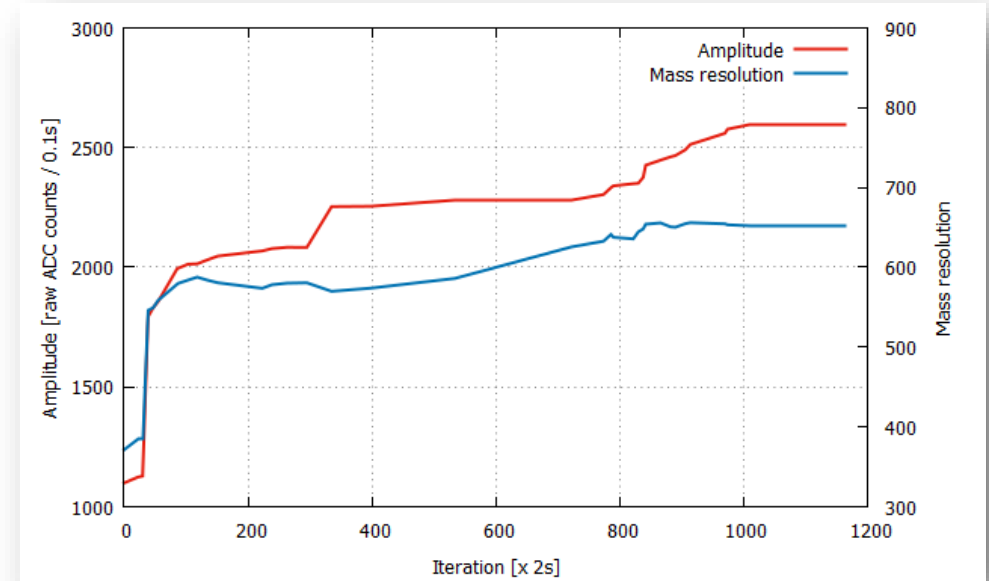


Real-Time Data Recording to Time Series Database (housekeeping and measurement results)

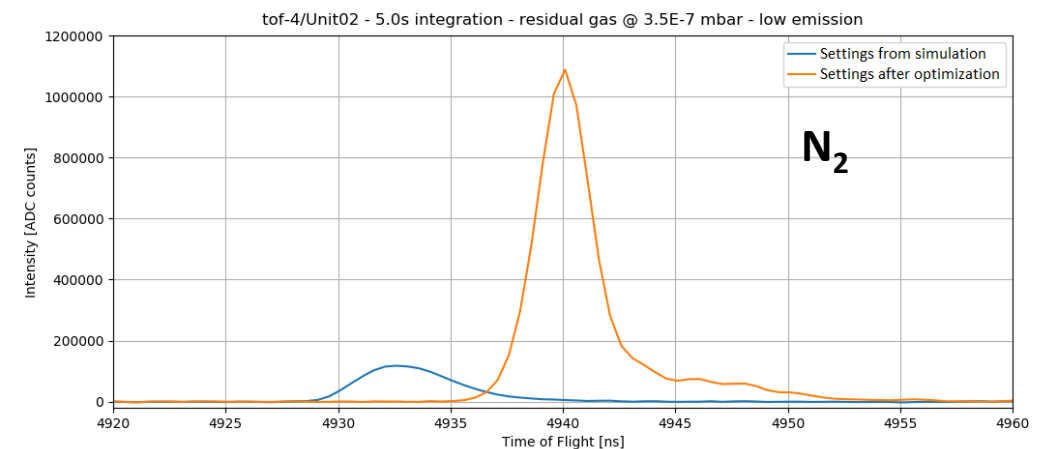


Automated Optimization

- Particle Swarm Algorithm (used on RTOF/Rosetta), which aim at finding the global optimum of ion optics settings (emitters, ion source, pulser, reflectron, etc.)
- Different score functions can be implemented: Area, Area/FWHM, Area/FWHM², Area²/FWHM, ...
- Uses residual gas peaks (e.g., CO₂, H₂O, ...), >10⁻⁷ mbar
- Performance: 1 optimization step every 2s



Example of high-voltages variation during optimization run

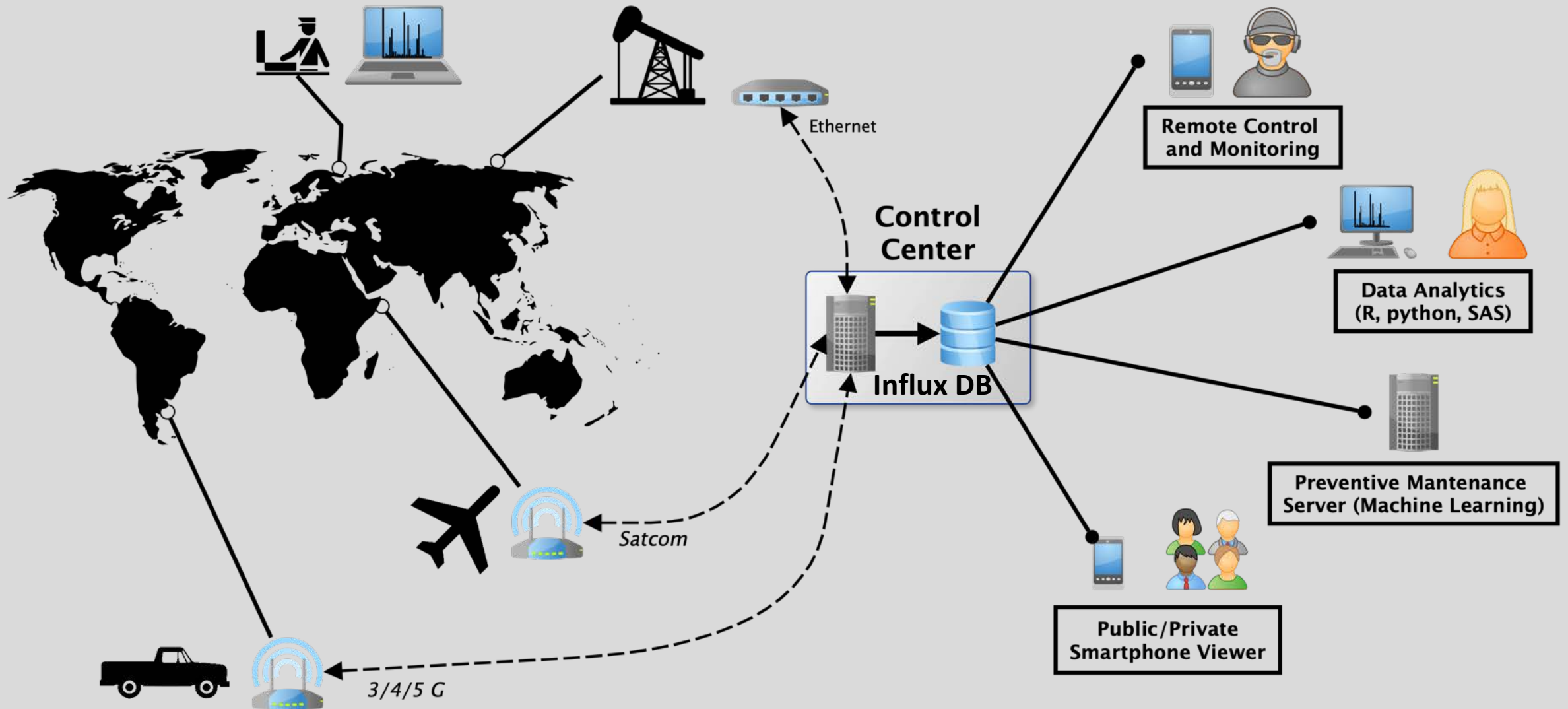




Vision

An IoT instrument enabling systems whose potential go beyond narrow definitions of capabilities of isolated 'offline' devices

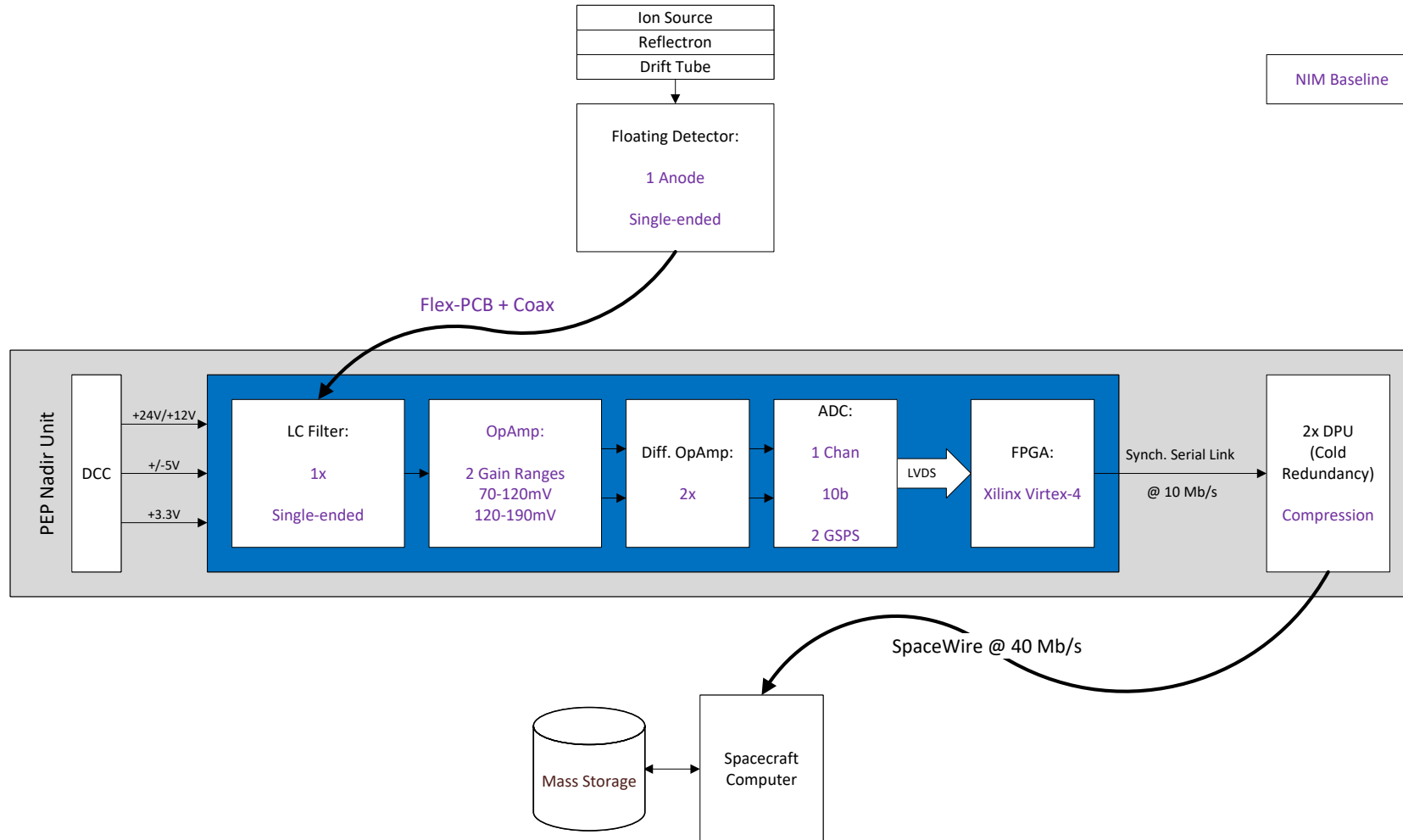
New ConOps



Thank You!

Looking forward to your questions, feedback, ideas, ...

Signal Chain of NIM

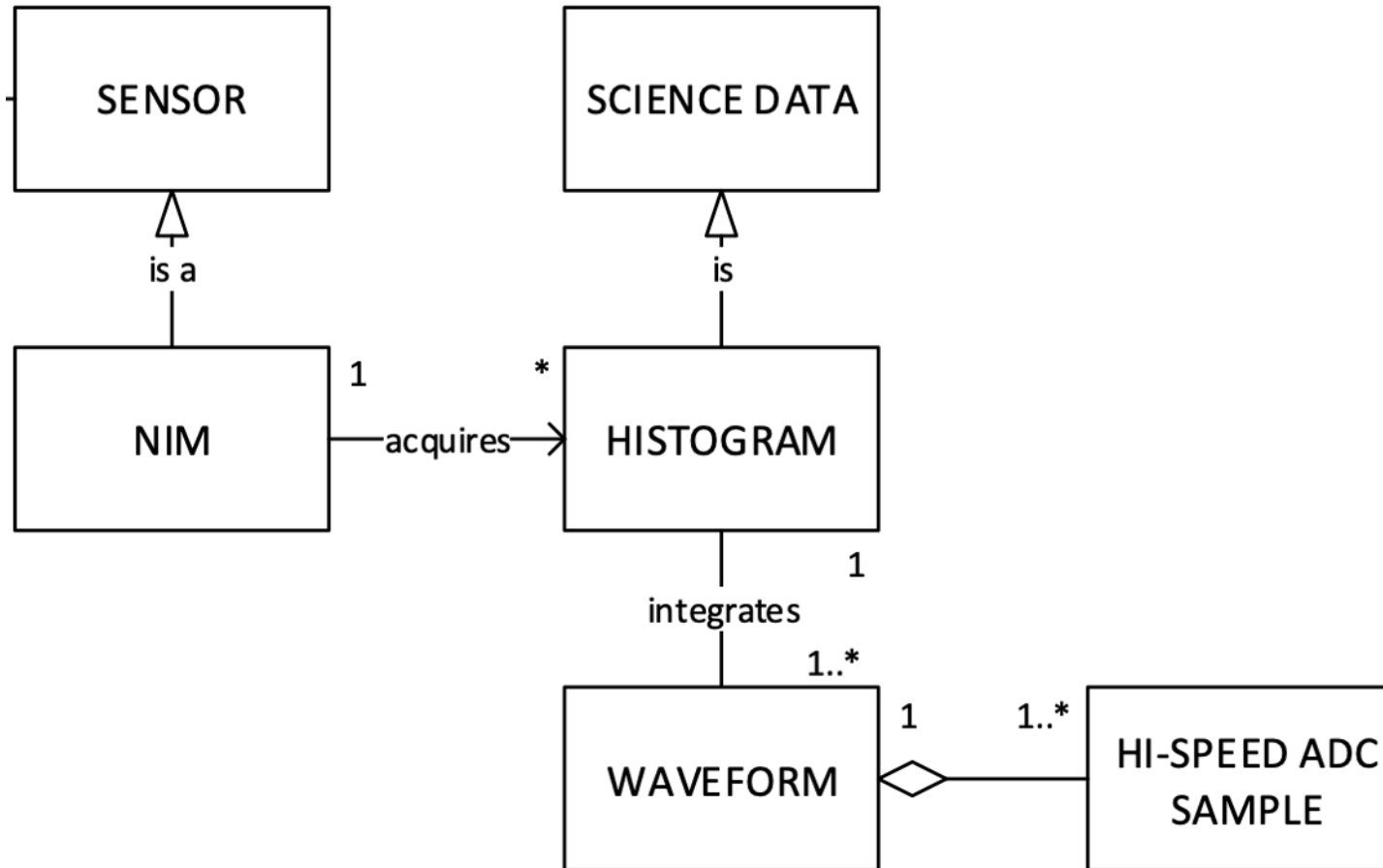


NIM Baseline

Key features:

- Floating detector
- 10 bit ADC, 2 GSPS
- LVDS link to FGPA (Virtex-4)

Data Processing of NIM



MG, v0.5

Expected Performance & Capabilities

From our space instrumentation experience:

- Speed: 0.1s (or longer, per spectrum)
- Mass resolution: $\sim 900 M/\Delta M$
- Mass range: 1-1000 u
- Dynamic range: 10^6 ($\sim 10s$ @ 10 kHz, 1 mV rms white noise)
- Detection level: 10^{-16} mbar

